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PTO/SB/16 (08-03)

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This is a request for filing a PROVISIONAL APPLICATION FOR PATENT under 37 CFR 1.53(c).

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123103

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Given Name (first and middle [if any])		Family Name or Surname		Residence (City and either State or Foreign Country)	
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Additional inventors are being named on the _____ separately numbered sheets attached hereto					
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ENCLOSED APPLICATION PARTS (check all that apply)					
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METHOD OF PAYMENT OF FILING FEES FOR THIS PROVISIONAL APPLICATION FOR PATENT					
<input checked="" type="checkbox"/> Applicant claims small entity status. See 37 CFR 1.27.					FILING FEE Amount (\$)
<input checked="" type="checkbox"/> A check or money order is enclosed to cover the filing fees.					80.00
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[Page 1 of 2]

Respectfully submitted,

SIGNATURE 

TYPED or PRINTED NAME Sean T. Bradley

TELEPHONE 913-339-9666

Date December 31, 2003

REGISTRATION NO. 46,572

(if appropriate)

Docket Number: 3350

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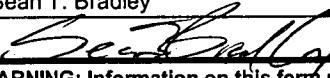
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Application Number	
Filing Date	
First Named Inventor	DONALD E. SCOTT
Examiner Name	
Group Art Unit	
Attorney Docket No.	3350

METHOD OF PAYMENT				FEE CALCULATION (continued)																																																																																																																																											
<p>1. <input checked="" type="checkbox"/> The Commissioner is hereby authorized to charge indicated fees and credit any overpayments to:</p> <p>Deposit Account Number 03-1425</p> <p>Deposit Account Name Chase & Yakimo</p> <p><input checked="" type="checkbox"/> Charge Any Additional Fee Required Under 37 CFR 1.16 and 1.17</p> <p><input checked="" type="checkbox"/> Applicant claims small entity status. See 37 CFR 1.27</p>				<p>3. ADDITIONAL FEES</p> <table border="1"> <thead> <tr> <th>Large Entity Fee Code (\$)</th> <th>Small Entity Fee Code (\$)</th> <th>Fee Description</th> <th>Fee Paid</th> </tr> </thead> <tbody> <tr><td>105</td><td>130</td><td>205</td><td>65 Surcharge - late filing fee or oath</td></tr> <tr><td>127</td><td>50</td><td>227</td><td>25 Surcharge - late provisional filing fee or cover sheet</td></tr> <tr><td>139</td><td>130</td><td>139</td><td>130 Non-English specification</td></tr> <tr><td>147</td><td>2,520</td><td>147</td><td>2,520 For filing a request for ex parte reexamination</td></tr> <tr><td>112</td><td>920*</td><td>112</td><td>920* Requesting publication of SIR prior to Examiner action</td></tr> <tr><td>113</td><td>1,840*</td><td>113</td><td>1,840* Requesting publication of SIR after Examiner action</td></tr> <tr><td>115</td><td>110</td><td>215</td><td>55 Extension for reply within first month</td></tr> <tr><td>116</td><td>400</td><td>216</td><td>200 Extension for reply within second month</td></tr> <tr><td>117</td><td>920</td><td>217</td><td>460 Extension for reply within third month</td></tr> <tr><td>118</td><td>1,440</td><td>218</td><td>720 Extension for reply within fourth month</td></tr> <tr><td>128</td><td>1,960</td><td>228</td><td>980 Extension for reply within fifth month</td></tr> <tr><td>119</td><td>320</td><td>219</td><td>160 Notice of Appeal</td></tr> <tr><td>120</td><td>320</td><td>220</td><td>160 Filing a brief in support of an appeal</td></tr> <tr><td>121</td><td>280</td><td>221</td><td>140 Request for oral hearing</td></tr> <tr><td>138</td><td>1,510</td><td>138</td><td>1,510 Petition to institute a public use proceeding</td></tr> <tr><td>140</td><td>110</td><td>240</td><td>55 Petition to revive - unavoidable</td></tr> <tr><td>141</td><td>1,280</td><td>241</td><td>640 Petition to revive - unintentional</td></tr> <tr><td>142</td><td>1,280</td><td>242</td><td>640 Utility issue fee (or reissue)</td></tr> <tr><td>143</td><td>460</td><td>243</td><td>230 Design issue fee</td></tr> <tr><td>144</td><td>620</td><td>244</td><td>310 Plant issue fee</td></tr> <tr><td>122</td><td>130</td><td>122</td><td>130 Petitions to the Commissioner</td></tr> <tr><td>123</td><td>50</td><td>123</td><td>50 Processing fee under 37 CFR 1.17(q)</td></tr> <tr><td>126</td><td>180</td><td>126</td><td>180 Submission of Information Disclosure Stmt</td></tr> <tr><td>581</td><td>40</td><td>581</td><td>40 Recording each patent assignment per property (times number of properties)</td></tr> <tr><td>146</td><td>740</td><td>246</td><td>370 Filing a submission after final rejection (37 CFR § 1.129(a))</td></tr> <tr><td>149</td><td>740</td><td>249</td><td>370 For each additional invention to be examined (37 CFR § 1.129(b))</td></tr> <tr><td>179</td><td>740</td><td>279</td><td>370 Request for Continued Examination (RCE)</td></tr> <tr><td>169</td><td>900</td><td>169</td><td>900 Request for expedited examination of a design application</td></tr> <tr><td colspan="4">Other fee (specify) _____</td></tr> <tr> <td colspan="4" style="text-align: right;">SUBTOTAL (1) (\$ 80.00)</td> <td colspan="4" style="text-align: right;">SUBTOTAL (3) (\$ -0-)</td> </tr> <tr> <td colspan="8"> <p>*or number previously paid, if greater; 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SUBMITTED BY		Complete if applicable		
Name (Print/Type)	Sean T. Bradley	Registration No. (Attorney/Agent)	46,572	Telephone 913-339-9666
Signature			Date	12/31/2003

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DIMENSIONALLY STABLE FABRIC

FIELD OF THE INVENTION

This invention relates to wool-synthetic blend fabrics and more particularly to flame-resistant, dimensionally stable wool-synthetic blend fabrics suitable for use in 5 aircraft interiors.

BACKGROUND OF THE INVENTION

Upholstery fabrics made from wool are known to have an attractive appearance and feel to the touch. Due to the tendency of wool to shrink after washing in water, however, attempts have been made to substitute wool fabrics with fabrics made from 10 synthetic materials such as polyester. The appearance and feel of fabrics made from synthetic materials, however, has been found to be inferior to that of fabrics made from wool. Fabrics made from blends of wool fibers with certain synthetic fibers retain some of the aesthetic features of wool as well as some of the cost benefits and potential property advantages of synthetics.

15 In the aircraft industry, seat cover fabrics are subject to specifications provided by aircraft manufacturers such as Airbus and Boeing. The relevant Airbus technical specification, for example, is TL 25/5092/83. The relevant flammability, smoke and toxicity portions of the standard are FAR 25.853 (b), appendix F, amended 32, JAR 25853 (b), appendix change 10, and ABD 0031 (previously numbered ATS 1000.001). These 20 specifications include standards for abrasion resistance including resistance to abrasion simulated by a Martindale tester. Resistance to stains resulting from spills, and to loss of color and shrinkage due to washing, is also specified. Seat cover fabrics may be required to meet specifications after a minimum of 10 washings. An areal weight below 470 g/m² is

(Docket 3350)

specified. It is desirable that shrinkage during service life, including shrinkage due to cleaning processes, be minimized. Resistance to pilling, corrosion and color loss may also be specified.

The relevant Boeing specification is BMS 8-236, for general upholstery interior applications. The flammability standard is provided by BSS7230, a twelve second vertical burn test, in which the sample is required to self extinguish within fifteen seconds, with a burn length of less than eight inches. Drips, if any, are required to extinguish in less than five seconds. Smoke emissions of less than 200 are specified according to BSS7238. Prescribed limits for individual toxic components in toxic gas emissions are tested according to BSS 7239. Dimensional stability is evaluated after prescribed cleaning, whether dry cleaning or water washing methods are used. While zero shrinkage is ideal, shrinkage levels of less than 6%, in both warp and fill directions, are acceptable. Standards for appearance, snag resistance, pilling resistance, color fastness and strength are part of the overall specification.

Wool fabrics are typically cleaned using a dry-cleaning process, including immersion in a solvent such as perchloroethylene, in order to maintain the dimensional stability of the fabric. Due to environmental and cost considerations, it would be desirable to clean wool-based fabrics without the use of perchloroethylene or other organic solvents. Water containing surfactants or detergents is highly effective in cleaning such fabrics, however, use of water-based cleaning solutions has been limited by the tendency of wool based fabrics to shrink after being subjected to such solutions. Synthetic fibers, on the other hand, are typically highly resistant to shrinkage following washing in water. Synthetic fibers, however, tend to be highly flammable.

Because of the nature of the constituent parts of the above mentioned wool-synthetic blends, such blends in the prior art are typically neither flame resistant, nor shrink resistant when washed in water. There is a need for fabrics made from wool-synthetic blends that will meet the special requirements for aircraft interiors.

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BRIEF DESCRIPTION OF THE INVENTION

In one embodiment of the invention, a method of producing a dimensionally stable, fire-resistant fabric suitable for use on aircraft includes the steps of providing a yarn a having a blend of wool fibers and fire-resistant synthetic fibers, the wool fibers comprising approximately 30% to 70% of the blend, weaving the yarn to form a fabric, and heat setting the fabric to achieve a washable woven structure resistant to shrinkage. The synthetic fibers may include polyester fibers produced or treated to enhance fire resistance.

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In another embodiment a method is provided for producing a dimensionally stable, fire-resistant fabric by spinning wool and fire-resistant polyester fibers to form a yarn, weaving the yarn to a form a fabric, and heat-setting the fabric to produce a finished material that passes Airbus and/or Boeing specifications.

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In a further embodiment a method is provided for producing a fire-resistant wool-based yarn by spinning shortened wool fibers with fire-resistant polyester fibers in a vortex spinning apparatus, weaving the yarn into an aircraft manufacturer-approved fabric, and heat-setting the fabric in a stenter apparatus such that it resists shrinkage after water washing. In a further embodiment, the method includes treating the yarn or fabric with zirconium to augment the fire-resistant properties.

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In yet another embodiment a method is provided for producing a dimensionally stable fabric by providing wool fibers, an effective percentage thereof cut or broken to fall within a selected length range, providing fire-resistant synthetic fibers, spinning the wool and synthetic fibers to produce a wool-synthetic blend yarn, wherein the wool fibers 5 comprise approximately 30% to 70% of the blend, weaving the yarn to form a fabric, and heat setting the fabric to produce a fabric that passes aircraft manufacturer specifications.

Wool fibers having a typical length of no greater than approximately five centimeters may be prepared by stretch-breaking. The synthetic fibers may include polyester fibers. Fibers may be spun by delivering the fibers to a ring spinning, air-jet spinning or 10 vortex spinning apparatus for spinning the fibers into a yarn. The fabric may be heat-set by securing and heating the fabric within a stenter. When passing the fabric through a stenter, sufficient heat is applied to set the fabric and produce a dimensionally stabilized fabric resistant to shrinkage. Further steps may include applying zirconium fire retardant to the fabric and applying a coating to bind the zirconium fire retardant to the fabric.

DETAILED DESCRIPTION

In one embodiment, wool fibers are first prepared by reducing their length.

Wool tops, consisting of fibers that are approximately 5.5 to 8 cm in length, are passed through a stretch-breaking machine to reduce their lengths to approximately 2 to 5 cm. It is 5 advantageous if the fibers are approximately 3 to 4 cm in length. It is advantageous if the wool fibers have diameters in the range of 17 to 25 microns, and particularly advantageous if the wool fibers have diameters in the range of approximately 22 to 25 microns.

After stretch breaking, the wool fibers are combined with flame retardant (FR) polyester fibers having a length of approximately 2 to 5 cm, and the resulting combined 10 fiber bundles are passed through one or more draw frames. The drafted wool and FR polyester fiber bundles are introduced into a spinning machine at such relative rates as to achieve wool contents in the range of approximately 30 to 70 percent. It is advantageous to the properties of the resulting fabric if the wool content is in the range of approximately 40 to 60 percent.

15 Spinning Technology

Typically, carding occurs prior to, or as an initial step in, the spinning process. Through carding, fibers are straightened and made relatively parallel to one another. After carding the fibers form a thin layer called a web. The web is gathered into a loose rope called a sliver. The sliver is typically wound into a large can and then moved to a draw 20 frame. In the drawing process, multiple cans of sliver are drawn together to form a combined sliver.

Ring spinning is a relatively slow spinning technology that typically yields a high quality yarn. During ring spinning, sliver is fed into the drafting zone of the ring spinning frame. The drafting zone has one roller that turns relatively slow and feeds the

sliver and another roller that turns relatively fast. The faster roller pulls out a few fibers at a time forming a fine stream of fibers that are fed to a rotating spindle inside a ring. As the spindle rotates, it drags a slower moving traveler on the ring. The ring twists the fibers as they are wound onto a bobbin that rides on the spindle. After spinning, the yarn may then be 5 used for weaving, perhaps after being further transferred to other holding structures. Ring spinning has been the preferred method of producing high quality wool yarns that demonstrate superior feel to the touch and abrasion resistance.

The air-jet spinning method uses air currents to twist fibers together, resulting in higher throughput and productivity than ring spinning. Air-jet spinning may be 10 used to spin blends of wool and FR polyester fibers, but yields yarns with reduced abrasion resistance in comparison with ring and vortex spinning.

The air vortex spinning method is a particularly efficient spinning method that is capable of spinning yarns at very high speeds and that yields a yarn having a relatively smooth texture and increased abrasion resistance. A vortex spinning apparatus 15 typically takes drawn sliver and drafts it to the desired yarn count via a four-roller drafting unit. The drafted fibers are then sucked into a nozzle where a high speed air vortex wraps the fibers around the outside of a hollow stationary spindle. Yarn twist is then imparted as the fibers are pulled down a shaft that runs through the middle of the spindle.

An example of a vortex spinning apparatus is described in the patent to 20 Mori, patent No. 6,370,858, hereby incorporated by reference. Mori discloses a vortex spinning method in which a drafted fiber bundle is supplied to a nozzle block and then to a hollow guide shaft. A core fiber is also fed to the nozzle block and then to the hollow guide shaft. Vortex air currents ejected from spinning nozzles in the nozzle block cause inversely turned fibers to wrap the fiber bundle and core fiber to create core yarn. The core fiber may

be multi-filament in which case the vortex air currents balloon the multiple filaments, resulting in the filaments being partially separated from one another. The vortex air currents insert the front ends of the fibers into the clearances between the separated filaments, and cause the other ends of the fibers to wrap around the multi-filament core fiber, resulting in the 5 creation of the core yarn.

The spinning speed of a vortex spinner is much faster than that provided by ring spinning with the ring method typically producing yarn at the rate of 20 meters per minute and the vortex method typically producing yarn at the rate of 400 meters per minute. The vortex method does not readily accommodate the longer fibers typically used in wool 10 spinning, however, and it has been found to be advantageous to reduce the fiber lengths as illustrated in the various embodiments of the invention disclosed herein.

Preparation of the Fabric

In the various embodiments contained herein, the spinning process used to produce the yarn may include ring spinning, air-jet spinning, air vortex spinning or other 15 appropriate means. It is advantageous, however, to spin the yarn using a vortex spinning method and apparatus.

After spinning, the yarn is typically dyed to a selected color and then woven into a fabric. The particular weave is typically determined by the requirements of the eventual use of the fabric. Appropriate weaves include those known for use by American 20 Airlines and United Airlines.

After weaving, the fabric is heat-set to increase dimensional stability of the fabric. It is advantageous if the heat setting includes the step of affixing the fabric within a stenter frame so that a given dimension may be controlled during the heat-setting process. The fabric is heat set within the stenter by heating the fabric to a temperature in excess of

100°C. The actual temperature used is primarily dependent upon the chemical nature of the synthetic fiber being used. Multiple heating bays may be used, each successive bay typically providing increased heat. In the case where a polyester fiber is used, the maximum temperature is typically set between approximately 170°C and 220°C. Dwell time, the time 5 period in which heat is applied to the fabric in the stenter may be adjusted according to temperatures used and composition of the fabric. The fabric is typically heated by provision of dry heat using appropriate means such as a gas fired burner and heat exchanger. In one embodiment, dimensional stability results from incipient melting of polyester (or other synthetic) fibers and subsequent bonding of the fibers to form a continuous or semi- 10 continuous polyester network or lattice within the fabric.

In an embodiment directed to vortex spun yarn, wool tops are passed through a stretch-breaking apparatus and the fiber length is thereby reduced to approximately 3 to 4 cm. The wool fibers are then combined with FR polyester staple having an approximate length of 3 cm, at a ratio of one part wool fiber to one part FR polyester. The 15 combined fibers are drafted on a drawframe and then spun in a vortex spinner. Portions of the yarn are dyed to a desired color or colors and then woven into a fabric suitable for use in aircraft such as for seat upholstery. The fabric is heat set in a stenter at approximately 190°C for approximately 30 seconds. As a result of this process the fabric meets airline interior fabric test specifications, including those for fire resistance, abrasion and shrinkage after 20 water washing. By way of example, a fabric may be produced in accordance with the above embodiment to pass Airbus specification TL 25/5092/83 and Boeing specification BMS 8-236.

Additionally, yarn spun from a blend of wool and synthetic fibers, the wool fibers comprising between approximately 30 to 70 percent of the blend, may be treated with

zirconium-based fire retardants prior to weaving to augment the fire-resistant qualities of the resulting fabric. The zirconium treatment may also be applied to any of the other fabrics set forth above.

To resist dislodging of the zirconium fire retardant from the fabric during

5 washing, the fabric may be treated with polyurethane or other appropriate material to coat the zirconium and bind it to the fabric.